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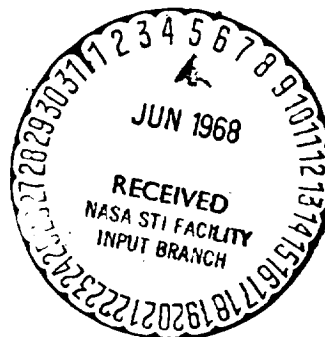
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# INVESTIGATION OF THE STABILITY OF A TWO-BLADED ROTOR ON AN ANISOTROPIC BASE<sup>†</sup>

R. S. Musalimova

(The article was presented by P. M. Riz, Professor  
at the Moscow Automechanics Institute.)

**ABSTRACT:** The results of the effect of the most important parameters (coefficient of anisotropy of the base  $S$ ; frequency of natural oscillations of the blade, caused by the flexibility in the vertical link  $P_{0b}$ ) on the stability are examined in this paper. In the only study [1] treating this problem, Coleman drew an invalid conclusion which stated that, if the rigidities of the base in different directions do not differ too greatly from one another ( $S \cong 1$ ), then we can consider the support isotropic. The results obtained here show the error in such a statement by Coleman.

## SOME RESULTS OF AN ANALYSIS OF THE STABILITY

The oscillations of a two-bladed rotor on an anisotropic base /31\*  
are described by the following equation:

$$\left. \begin{aligned} \ddot{x} + 2n_x \dot{x} + P_x^2 x + \epsilon_1 [(\ddot{\eta} - \omega^2 \eta) \sin \omega t + 2\omega \dot{\eta} \cos \omega t] &= 0, \\ \ddot{z} + 2n_z \dot{z} + P_z^2 z + \epsilon_1 [(\ddot{\eta} - \omega^2 \eta) \cos \omega t - 2\omega \dot{\eta} \sin \omega t] &= 0, \\ \ddot{\eta} + 2n_\eta \dot{\eta} + (\omega^2 \nu_0^2 + P_{0b}^2) \eta + 2\epsilon_2 [\ddot{x} \sin \omega t + \ddot{z} \cos \omega t] &= 0, \end{aligned} \right\} \quad (1)$$

where  $x$  and  $z$  are the shifts of the rotor center toward the direction of the axes (Fig. 1 in [2]),

$$\eta = \xi_1 - \xi_2,$$

$\xi_1$  and  $\xi_2$  are the angular deviations of the blades (Fig. 1 in [2]),

<sup>†</sup> See the beginning in No. 3, 1967.

\* Numbers in the margin indicate pagination in the foreign text.

$$P_x^2 = C_x/M, \quad P_z^2 = C_z/M,$$

$P_x$  and  $P_z$  are the frequencies of natural oscillations of the base toward the direction of the  $x$  and  $z$  axes,

$$2n_x = K_x/M, \quad 2n_z = K_z/M,$$

$C_x, C_z, K_x, K_z$  are the coefficients of rigidity and damping of the flexible base toward the direction of the  $x$  and  $z$  axes,

$$P_{0b}^2 = C_b/J_b, \quad 2n_b = K_b/J_b,$$

$P_{0b}$  is the frequency of natural oscillations of the blade, caused by the flexibility in the vertical link, /32

$C_b$  and  $K_b$  are the angular coefficients of rigidity and damping of the blade in the vertical link,

$J_b$  is the moment of inertia of the blade in relation to the vertical link,

$$\epsilon_1 = \frac{m_b r}{m + 2m_b},$$

$m_b$  is the mass of the blade,

$r$  is the radius of the center of gravity of the blade, in relation to the vertical link,

$M$  is the mass of the flexible base (reduced mass of the body),

$$v_0^2 = S_b l_{v1}/J_b,$$

$S_b$  is the static moment of the mass of the blade, in relation to the vertical link,

$l_{v1}$  is the loss of the vertical link,

$$\epsilon_2 = r/\rho^2,$$

$\rho$  is the radius of inertia for the blade, in relation to the vertical link.

Let us turn to the system in dimensionless coordinates. We will assume that  $\omega t = \psi$ .

$$\left. \begin{aligned} \frac{d^2 \bar{x}}{d\psi^2} + 2\bar{n}_x \frac{1}{\omega} \frac{d\bar{x}}{d\psi} + \frac{1}{\omega^2} \bar{x} + \bar{\epsilon}_1 \left[ \left( \frac{d^2 \eta}{d\psi^2} - \eta \right) \sin \psi + \right. \\ \left. + 2 \frac{d\eta}{d\psi} \cos \psi \right] = 0, \end{aligned} \right\} \quad (2)$$

$$\frac{d^2 \bar{z}}{d\psi^2} + 2\bar{n}_x \frac{1}{\omega} \frac{d\bar{z}}{d\psi} + S^2 \frac{1}{\omega^2} \bar{z} + \bar{\epsilon}_1 \left[ \left( \frac{d^2 \eta}{d\psi^2} - \eta \right) \cos \psi - \right. \\ \left. - 2 \frac{d\eta}{d\psi} \sin \psi \right] = 0,$$

(2 Cont.)

$$\frac{d^2 \eta}{d\psi^2} + 2\bar{n}_b \frac{1}{\omega} \frac{d\eta}{d\psi} + \left( \nu_0^2 + \bar{P}_0^2 \frac{1}{\omega^2} \right) \eta + \\ + 2\bar{\epsilon}_2 \left[ \frac{d^2 x}{d\psi^2} \sin \psi + \frac{d^2 \bar{z}}{d\psi^2} \cos \psi \right] = 0.$$

Here  $x = \bar{x}/r$ ;  $\bar{z} = z/r$ ;  $\bar{n}_x = n_x/P_x$ ;  $\bar{n}_z = n_z/P_x$ ;  $\bar{n}_b = n_b/P_x$ ;  $\bar{P}_{0b} = \bar{P}_{0b}/P_x$ ; and  $S = P_z/P_x$  is the coefficient of anisotropy:

$$\bar{\omega} = \frac{\omega}{P_x}; \quad \bar{\epsilon}_1 = \frac{m_A}{M + 2m_{Bx}}; \quad \bar{\epsilon}_2 = \left( \frac{r}{\rho} \right)^2; \quad \epsilon = \bar{\epsilon}_1 \bar{\epsilon}_2.$$

System (2) is equivalent to the following system:

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$$\begin{aligned} \dot{y}_1 &= y_4, \quad \dot{y}_2 = y_5, \quad \dot{y}_3 = y_6, \\ y_1 &= \frac{1}{1-2\epsilon} \left\{ \frac{1}{\omega^2} (2\epsilon \cos^2 \psi - 1) y_1 - \epsilon \sin 2\psi \frac{S^2}{\omega^2} y_3 + \right. \\ &+ \bar{\epsilon}_1 \sin \psi \left( 1 + \nu_0^2 + \frac{\bar{P}_{0A}^2}{\omega^2} \right) y_2 + 2\bar{n}_x \frac{1}{\omega} (2\epsilon \cos^2 \psi - 1) y_4 - \\ &- 2\bar{n}_z \frac{1}{\omega} \epsilon \sin 2\psi y_5 - 2\bar{\epsilon}_1 \left[ (1-2\epsilon) \cos \psi - \bar{n}_x \frac{1}{\omega} \sin \psi \right] y_6 \Big\}, \\ y_2 &= \frac{1}{1-2\epsilon} \left\{ -\frac{1}{\omega^2} \epsilon \sin 2\psi y_1 - \frac{S^2}{\omega^2} (1-2\epsilon \sin^2 \psi) y_2 + \right. \\ &+ \bar{\epsilon}_1 \cos \psi \left( 1 + \nu_0^2 + \frac{\bar{P}_{0A}^2}{\omega^2} \right) y_3 - 2\bar{n}_x \frac{\epsilon}{\omega} \sin 2\psi y_4 - \\ &- 2\bar{n}_z \frac{1}{\omega} (1-2\epsilon \sin^2 \psi) y_5 + 2\bar{\epsilon}_1 \left[ (1-2\epsilon) \sin \psi + \bar{n}_x \frac{1}{\omega} \cos \psi \right] y_6 \Big\}, \\ y_3 &= \frac{1}{1-2\epsilon} \left\{ 2\bar{\epsilon}_2 \frac{1}{\omega^2} \sin \psi y_1 + 2\bar{\epsilon}_2 \frac{S^2}{\omega^2} \cos \psi y_2 - \left( \nu_0^2 + 2\epsilon + \frac{\bar{P}_{0A}^2}{\omega^2} \right) \times \right. \\ &\times y_3 + 4\bar{\epsilon}_2 \bar{n}_x \frac{1}{\omega} \sin \psi y_4 + 4\bar{\epsilon}_2 \bar{n}_z \frac{1}{\omega} \cos \psi y_5 - 2\frac{\bar{n}_A}{\omega} y_6 \Big\}, \\ y_1 &= \bar{x}, \quad y_2 = \bar{z}, \quad y_3 = \eta, \quad y_4 = d\bar{x}/d\psi, \quad y_5 = d\bar{z}/d\psi, \quad y_6 = d\eta/d\psi. \end{aligned} \quad (3)$$

System (3) examines the stability according to the method mentioned in a previous article [2]. All the calculations were made

on an electron computer M-20. For this purpose, we made up a special program.

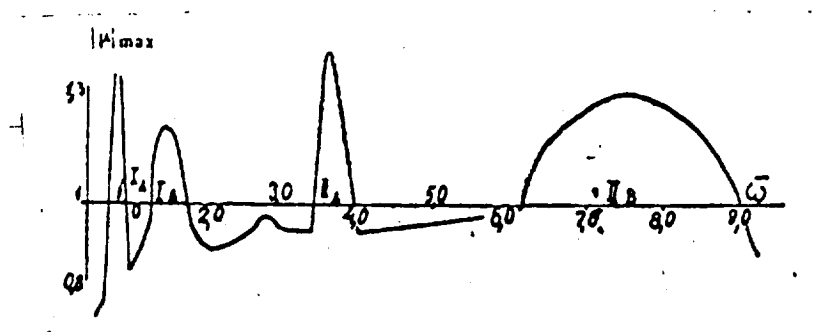


Fig. 1. Graph of the Dependence of  $|\mu|_{\max}$  on the Relative Convolutions of the Rotor  $\bar{\omega} = \omega/D_x$ ,  $S = 4$ .

The behavior of  $|\mu|_{\max}$  within the range for a change in the relative convolutions of a rotor  $\bar{\omega} \in (0.5; 9.1)$ , with  $S = 4$ , is shown in Figure 1. There are two pairs of ranges (zones) of instability. We will call the first pair of zones  $I_A$  and  $I_B$ , and the second pair  $II_A$  and  $II_B$ . If we are interested in only one pair, then we will call the zones A and B. In the  $I_A$  and  $II_A$  zones it was found that  $Jm\mu = 0$ . These zones will be called aperiodic instability zones. In the  $I_B$  and  $II_B$  zones,  $Jm\mu \neq 0$ . These zones will be called oscillational instability zones. In the  $I_A$  and  $I_B$  zones, oscillations occur along the  $x$ -axis. In the  $II_A$  and  $II_B$  zones, they occur along the  $z$ -axis. /34

(a) Effect of the Anisotropy Coefficient of the Base on the Stability. We are examining the effect of  $S$  on the instability zones during a decrease in  $S$  from 3 to 1. It was assumed that, as  $S$  approaches 1, the two pairs of zones above will converge or, more accurately, the second pair of zones ( $II_A$  and  $II_B$ ) will approach the first pair ( $I_A$  and  $I_B$ ), and will combine with it when  $S = 1$ . But it was found that, with values of  $S$  close to 1, the picture becomes more complicated. A certain additional zone  $D$  makes its appearance. There comes a moment when, in the interval of practical convolutions of the rotor  $\bar{\omega} \in (0; 5; 3.0)$ , there are five entire, fairly expansive, instability zones ( $S = 1.4$ ), Figure 2.

Zone D occurs at  $S \approx 2.2$ , isolated from the right of the  $II_A$  zone. With a further decrease in  $S$ , the D zone breaks completely from the  $II_A$  zone. Thus  $Jm\mu \neq 0$  in the D zone. Therefore, we will also call it an oscillational instability zone. While the  $II_A$  and  $I_B$  zones approach the  $I_A$  zone, as we would anticipate, the height  $H$  and the width  $\Delta\bar{\omega}$  of these zones ( $II_A$  and  $I_B$ ) decrease significantly; but the  $II_A$  zones increase, as we can also see from the Table (the lower number in the boxes is the value of  $\Delta\omega$  multiplied by  $10^2$ , and the upper number is the value of  $H$ , multiplied by  $10^4$ ).

Thus, we achieve the greatest  $H$  (zone A-7766, zone B-2468) and width of the zone  $\Delta\bar{\omega}$  (zone A - 28, zone B - 90) when  $S = 1$ .

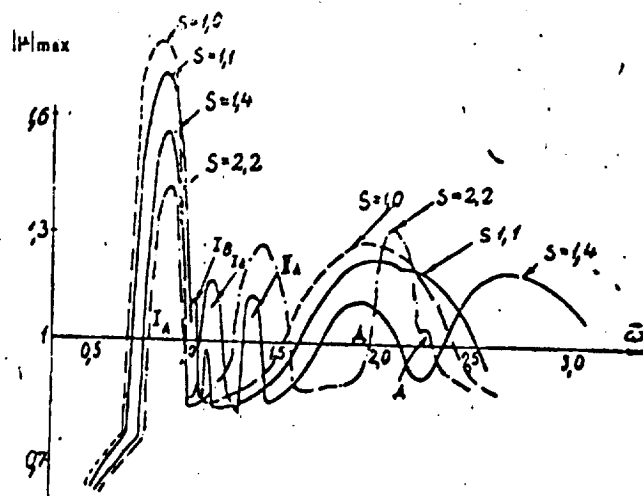


Fig. 2. Graph of the Greatest Multiplier (by Modulus) of  $|\mu|_{\max}$  Versus the Anisotropy Coefficient for a Value of  $S = P_z/P_x$  Close to 1.  $\pi_x = \pi_z = n_b = 0.05$ ;  $v_0 = 0.3$ ;  $\epsilon = 0.05$ ;  $\bar{P}_{0b} = 0$ ;  $\bar{\epsilon}_2 = 0.894$ ;  $\Delta\psi = 2\pi/100 = 3.6^\circ$ .

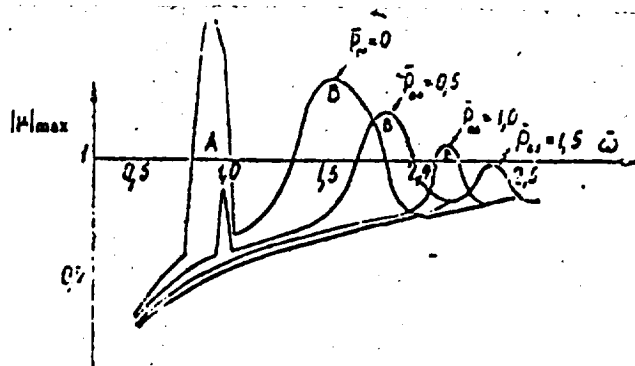


Fig. 3. Graph of the Greatest Multiplier (by Modulus) of  $|\mu|_{\max}$  Versus the Relative Natural Frequencies of the Blade in the Vertical Link  $\bar{P}_{0b}$ .  $\epsilon = 0.05$ ;  $\pi_x = \pi_z = n_b = 0.05$ ;  $v_0 = 0.3$ ;  $\bar{\epsilon}_2 = 0.894$ ;  $S = 4$ .

(b) Effect of Relative Frequency of Natural Oscillations of a Blade, Caused by the Flexibility in the Vertical Link  $\bar{P}_{0b}$ , on the Stability (Fig. 3). With an insignificant increase in  $\bar{P}_{0b}$ , zone A disappears. In this case, B shifts to the right, and the amplitude and width of zone B decrease significantly. When  $\bar{P}_{0b}$  is

equal to 1.5, zones A and B disappear completely. These calculations were made for damping coefficients of  $\bar{n}_x = \bar{n}_z = \bar{n}_b = 0.05$ .

$\frac{S}{H}$	3	1.8	1.2	1
$I_A$	3735 18	4402 20	6767 24	7766 28
$I_B$	2261 40	2201 20	1241 4	0 0
$II_A$	3592 19	2235 18	721 7	0 0
$\Delta$	— —	722 38	1446 46	
$II_B$	— —	— —	1683 74	

#### REFERENCES

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